	Copy Approved for Release 2010/09  SECURITY INF  CENTRAL INTELL  INFORMATIC		REPORT
COUNTRY	JSSR	. * *	DATE DISTR. Q JULY 52
SUBJECT			NO. OF PAGES 5 25X1
	Guided Miss	ile Development	
PLACE ACQUIRED			NO. OF ENCLS.
DATE ACQUIRED			SUPPLEMENT TO REPORT NO.
DATE OF			25X1
OF THE UNITED STATES, WI AND 784, OF THE U.S. COL LATION OF ITS CONTENTS	INFORMATION AFFECTING THE NATIONAL DEFENSE ITHIN THE MEANING OF TITLE 18, SECTIONS 793 DE, AS AMENDED. ITS TRANSMISSION ON REVE- TO OR RECEIT BY AN UNAUTHORIZED PERSON IS REPRODUCTION OF THIS FORM IS FRONTSITED.	THIS IS UNEV	ALUATED INFORMATION
Append	lix <sup>0</sup> B <sup>8</sup>		
Traviani associated	Guided Missiles - See separa	te sheets attached.	25X1
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25X1



## SECURITY INFORMATION 25X1 SECRET -3-Appendix 'B' Page 1 GUIDED MISSILES 25X1 expansion ratio (exit area to throat area) of the engine <u>(a)</u> 25X1 Model Exp. Ratio Comb. Press. Theoretical Exhaust Velocity 2000 m/sec. 25 ton 15 atm. 2000 m/sec. 35 ton 20 atm. 16.0 60 atm. 100 ton 2360 m/sec. The higher expansion ratio for the 100 ton model follows from the greater combustion pressure employed. 25X1 (b) Model Max. Diam. Throat a Exit Comb. Chamber Diameter Diameter ton 400 mm. 740 mm. 35 950 " 400 740 mm. 470 1200 1800 The above are inside dimensions. The outside diameter would exceed the above by twice the depth of the cooling jacket, i.e. by about 30-50 mm. 25X1 (c) The standard A-4 has a specific thrust of 210 Kg. If this figure is maintained for the larger (100 ton) model, the designed fuel consumption of 500 Kg/sec. would yield 105 tons thrust. With the paraffin ruel, a higher specific thrust 25X1 is expected. 25X1 (<u>d)</u> The theoretical value for paraffin is 240 kg., giving a maximum theoretical thrust of 120 tons for a fuel consumption of 500 Kg/sec. (e) 25X1 (x)The paraffin has the characteristics of ordinary Kerosene (lamp oil). The specific impulse is 240 Kg. (g) The single stage centrifugal pumps work against a head of about 2300 ft. Pumps of known design can provide about 600-800 ft. head at high flow. It is admitted that the fuel pump (alcohol) on the standard A-4 has a delivery pressure of 25 atmospheres ( 800 ft.) at 3800 rpm. The same pump on the 35 ton model running at 5000 rpm., reaches a pressure of 35 atmospheres ( 1100 ft.). These are test figures for single stage, single entry centrifugal pumps. With double entry pumps higher delivery 25X1 are anticipated and with minor alterations in blade design the 70 atmosphere delivery pressure should be

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realizable.

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(h)		
- [	The nozzle shape was based on theoretical considerations.	8
( <u>i)</u>	Calculations show that a Hydrocarbon such as "paraffin" may be used successfully to regeneratively cool an engine of this size.	
l	The possibility of ultimately using the cooling jacket as a steam generator to assist the peroxide plant was considered.	
(1)		
	The original intention was to vapourize the water, and subsequently condense it by means of liquid oxygen.	
<u>k)</u>		
	In the final design, the water was not thrown away.	
1)		
<u>.</u>		
-	several cooling alternatives were investigated	
	it is quite possible that in the final version alcohol will replace the paraffin, and the normal A-4 cooling process will be adopted.	
m)	The water pump is reportedly "carried" by the paraffin pump.	
	The pumps are mounted on either side of the steam turbine, with the Oxygen and Peroxide pumps on the left and the paraffin and water pumps on the right. The casings of the individual pumps are located on each other, by means of special pads and the drive is by a common sub-divided shaft.	
<u>n)</u>	For the flow rate of Hydrogen Peroxide, the calculated turbine efficiency is 55%.	
	The 75% efficiency was hoped for in the final design.	
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<u>o)</u>	and lyb continuous, and makes and am time among continuous	
<u>o)</u>		

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whiteh could nouse the pump assembly	igned
whiten could nouse the pump assembly a thrust frame was desi	igned
	1
(r) In the second chamber design, a second flat copper sheet was welded to the side of the corrugated copper plate.	he innon
	6
(g)	25X
outer steel shell is 5 mm thick, and is required for strength.	The 25X
The liner consisted of pure copper (commercial grade). In version (1)	2.7.
ote: The turbo-pump assembly of the standard V.2 weights 180 Kg.	25X
The estimated design weight of the complete 2-stage steam turbine assem 350 Kg. Other component weights not known.	bly is
Paraffin " 360 mm rotor Water " 210 mm rotor Peroxide " 280 mm rotor	
Oxygen pump	, <b>, , , , , , , , , , , , , , , , , , </b>
The weight of the 100 tons combustion chamber together with the exhause is 800 Kg. The turbo-pump unit consists of the 2-stage steam turbing (	25X
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